



# **GLM Instrument Performance Verification**

**GLM Science Meeting  
September 27-29, 2016**

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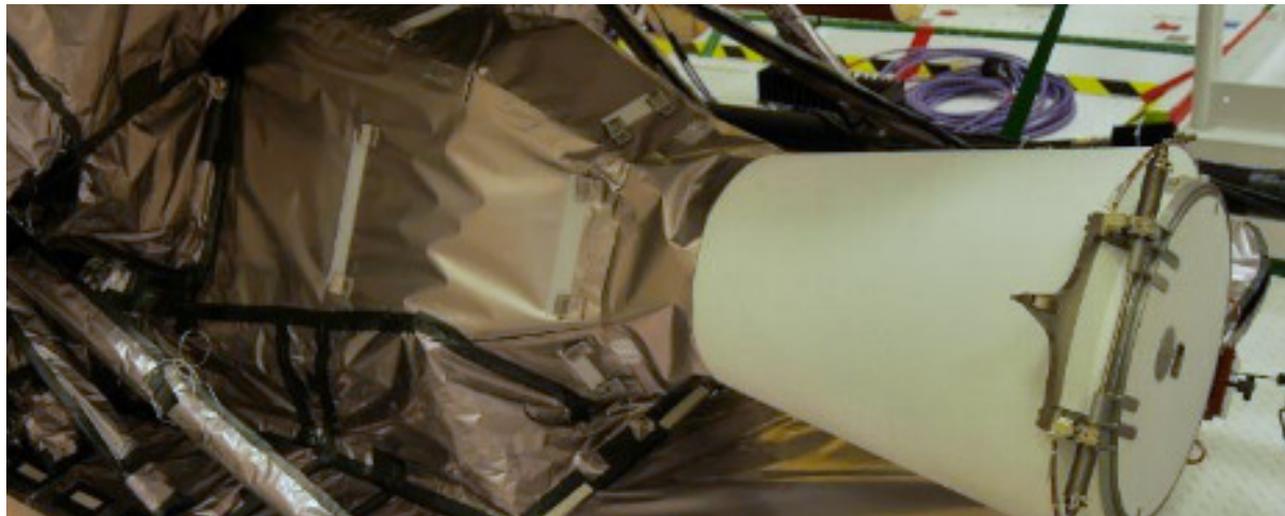
Lockheed Martin, Advanced Technology Center



# GLM Key Performance Requirements

Performance Requirement	GOES-R Capability
Detection efficiency > 70%, averaged over full disk and 24 hours.	> 70%
Coverage required over full disk	Meets coverage requirements
Flash false alarm rate shall be less than 5%, averaged over 24 hours	< 5%
Dynamic range greater than 100 at all times everywhere in FOV	< 100
Navigation error within $\pm 112$ microradians ( $\sim 1/2$ pixel or $\sim 4$ km)	< 112 $\mu$ rad

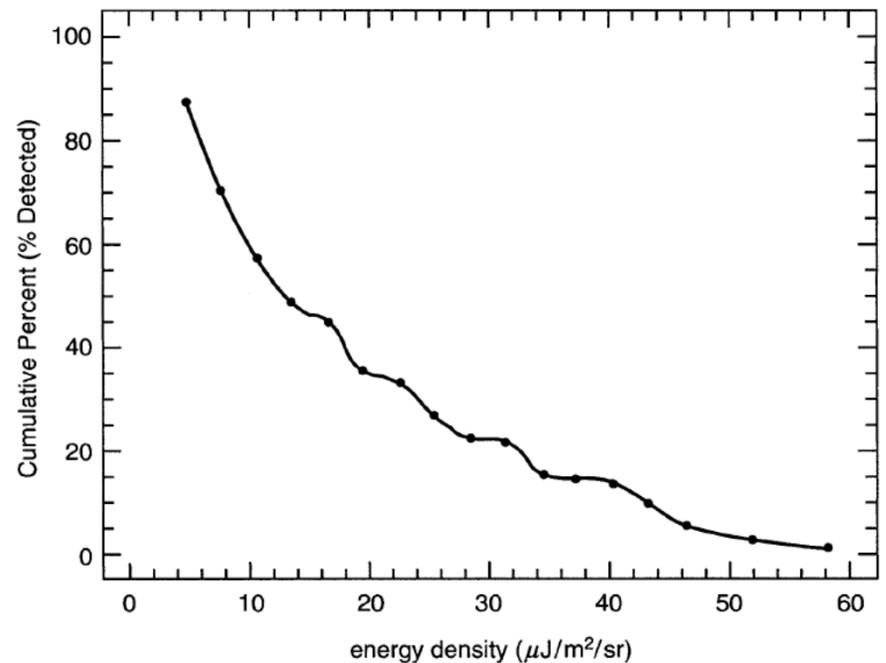
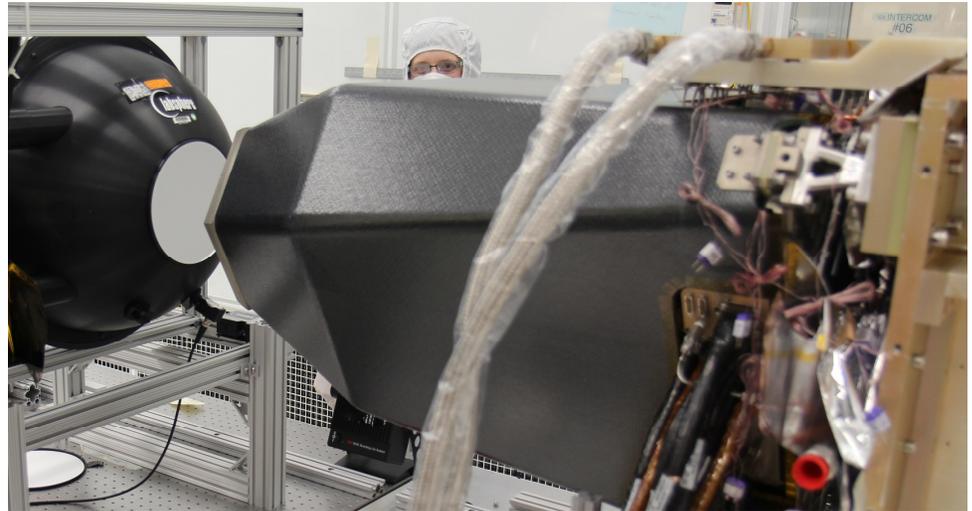
*GLM Flight Model 1  
Installed on GOES-R*





# Detection Efficiency

- Verification followed the example set by TRMM LIS
- Static and transient response of the instrument was measured using a calibrated source
- RTEP thresholds were set with a conservative false event rate
- Threshold levels were compared to U2 lightning radiance data to determine detection efficiency
- Final operational detection efficiency will depend on RTEP thresholds selected during PLT

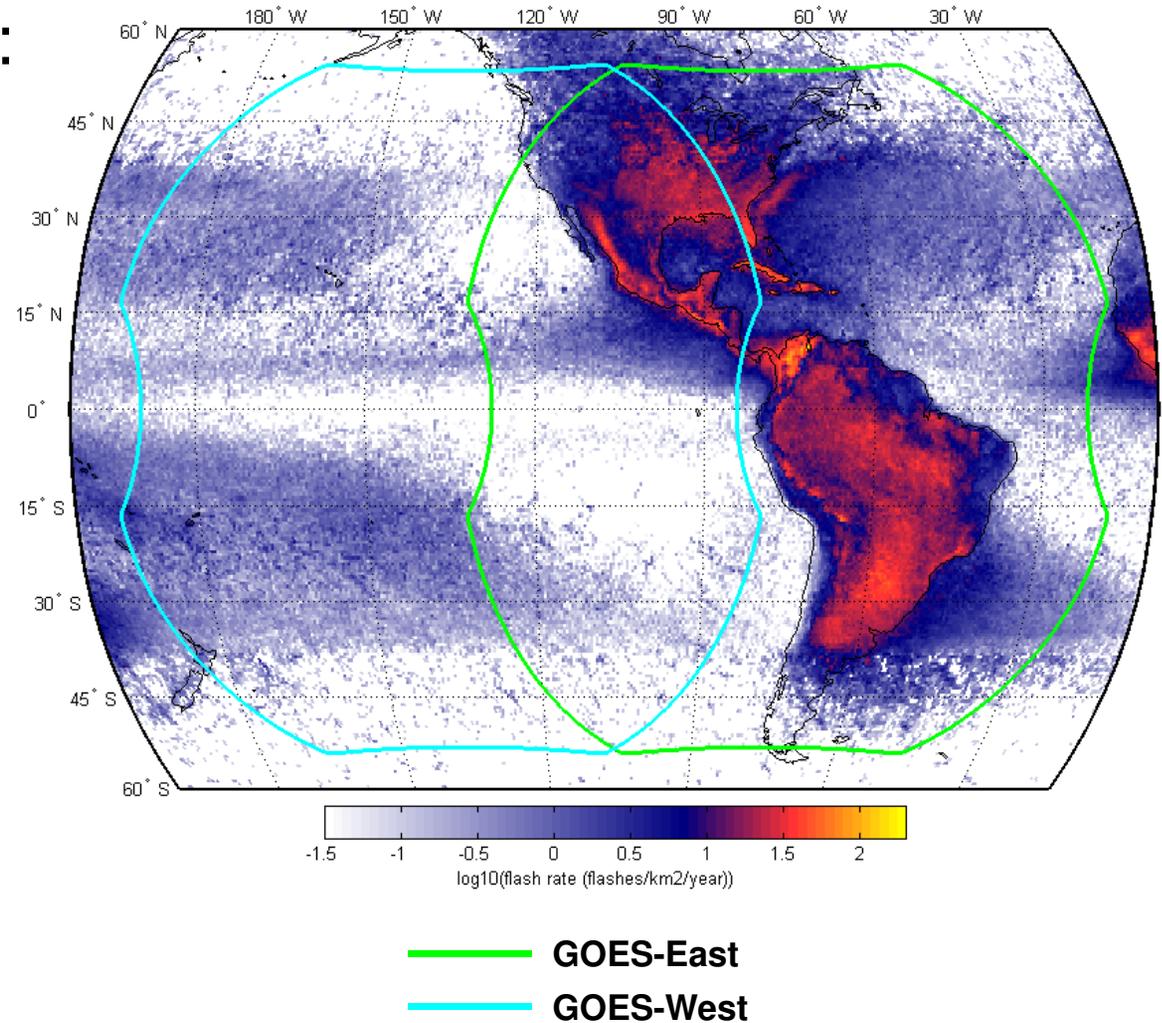


*Cumulative distribution of lightning optical energies, derived from U2 aircraft measurements. Used to determine DE for LIS and GLM.*



# Coverage

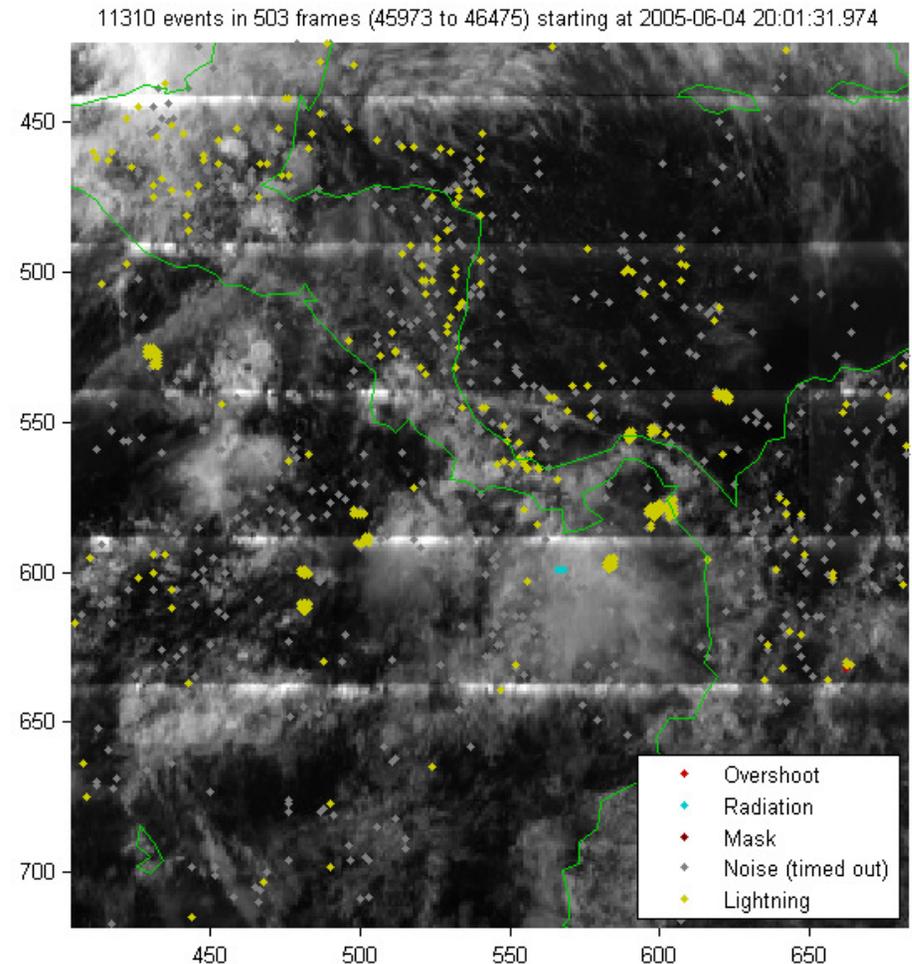
- Coverage is verified with a combination of:
  - Measured instrument placement on the spacecraft Earth Pointing Platform
  - Measured internal instrument alignment
  - Measured instrument Field of View





# False Alarm Rate

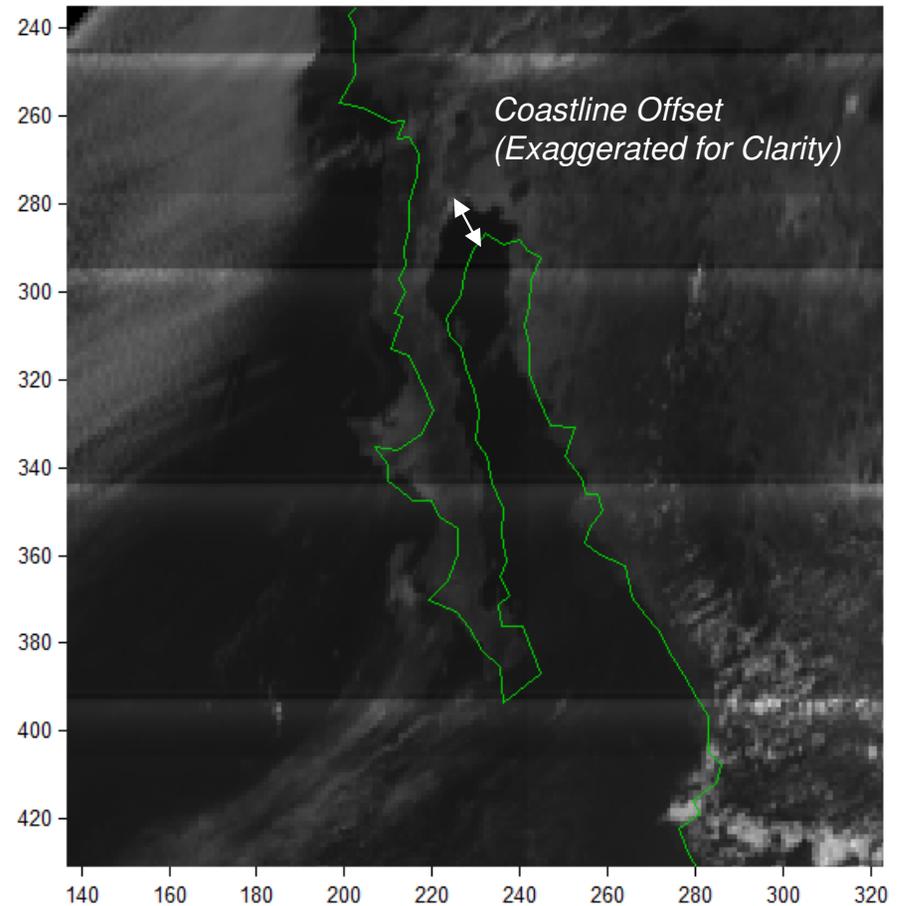
- False alarm rate is defined as the number of false flash detections divided by the average true flash rate
- Verification is by analysis
  - LM implementation of the GPAs is run on simulated and test data sets





# Navigation

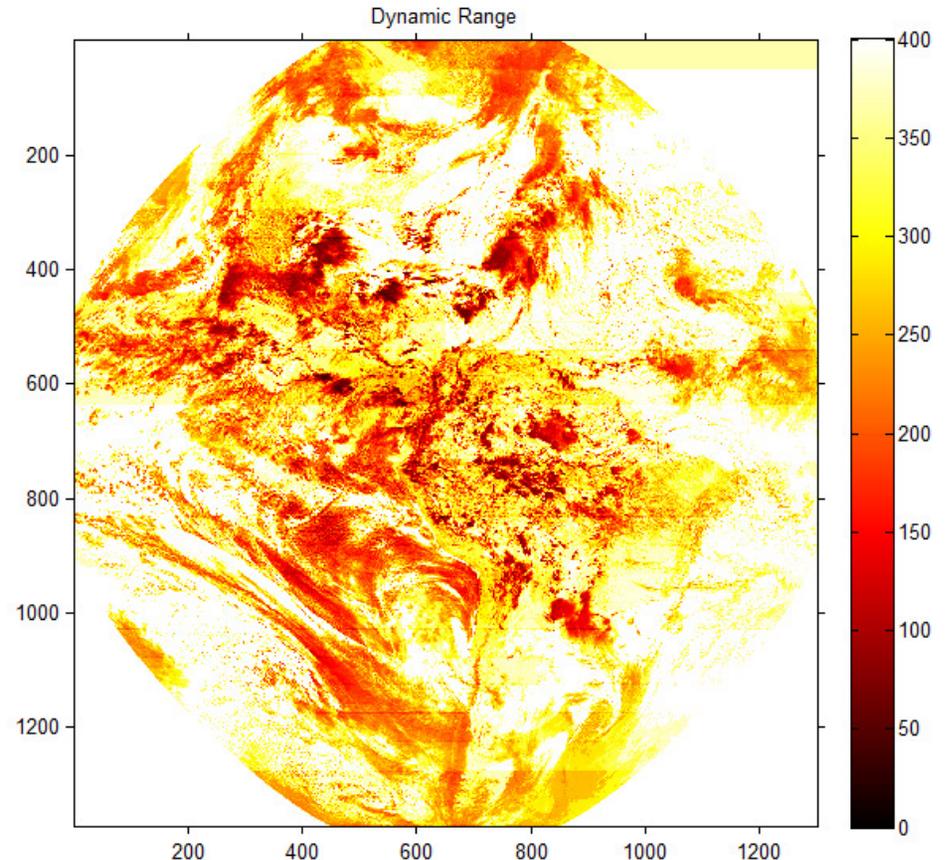
- GLM has no internal pointing mechanisms, and relies wholly on the spacecraft for pointing
- We register background images to the coastline database to determine offset between GLM boresight and spacecraft pointing
- We measure the optical distortion to determine pixel to pointing vector transformation
- We use analysis and test to predict internal GLM pointing response to expected temperature distributions





# Dynamic Range

- Dynamic Range is defined as the ratio of available headroom above background (under worst case illumination conditions) to the lowest detectable signal (RTEP threshold)
- Due to higher than designed responsivity of the instrument, GLM does not meet the dynamic range requirement at certain times of the day and regions of the FOV
- Impact on Detection Efficiency and False Alarm Rate is expected to be minor, based on simulation of typical cloud background scenes (based on current GOES images)
  - Small percentage of events may be saturated, a condition that is readily detectable





# **GLM Post Launch Test (PLT) “Vendor” Tools**

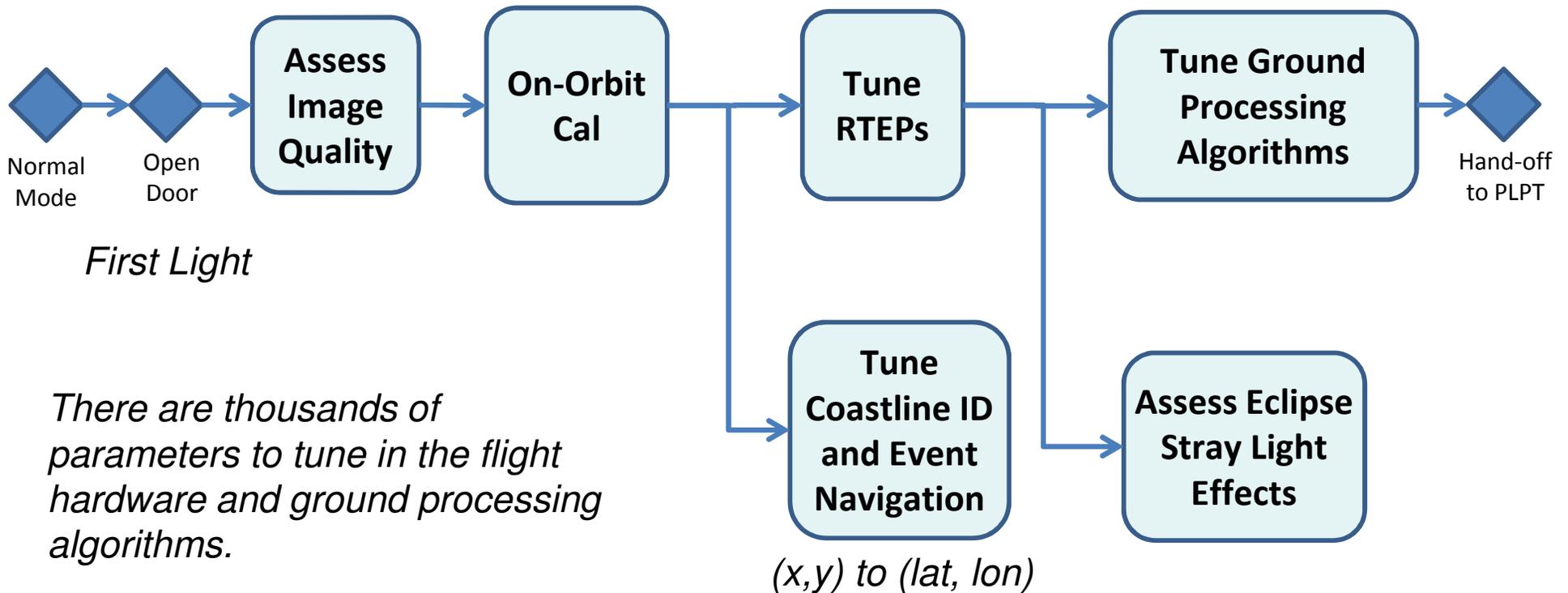
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# GLM Post Launch Test Flow

**Follows Signal Chain:** Electrons → DN    DN → Events (downlink)    Events → Level 1b Lightning



*First Light*

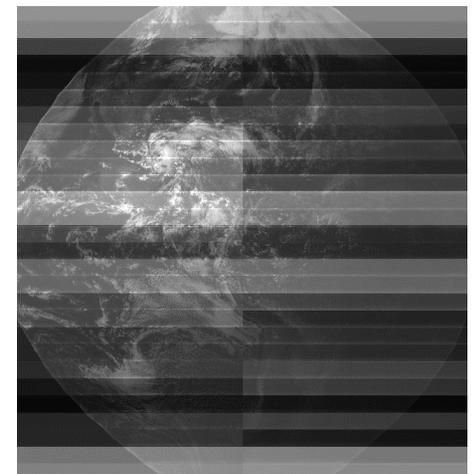
*There are thousands of parameters to tune in the flight hardware and ground processing algorithms.*

*Good tools are essential.*



# Image Quality

- Video is acquired at 503 fps
- RTEPs maintain an exponential average of the value of each pixel
- Periodically, GLM collects a “snapshot” of this background average
  - In normal mode, this occurs every 150 seconds
  - In “fast background” mode (with the entire science telemetry stream dedicated to background, and no events) this occurs every 12 seconds
- Matlab PLT tool is used to extract image metrics, such as...
  - Cross-cuts, Headroom to ADC saturation, Contrast, Overshoot, Dynamic Range, Estimated gain ( $\text{DN}/(\text{mW}/\text{cm}^2\text{-sr-}\mu\text{m})$ ), Max, Min, Zones of Reduced Data Quality, Down-sampled images, Histograms, etc.

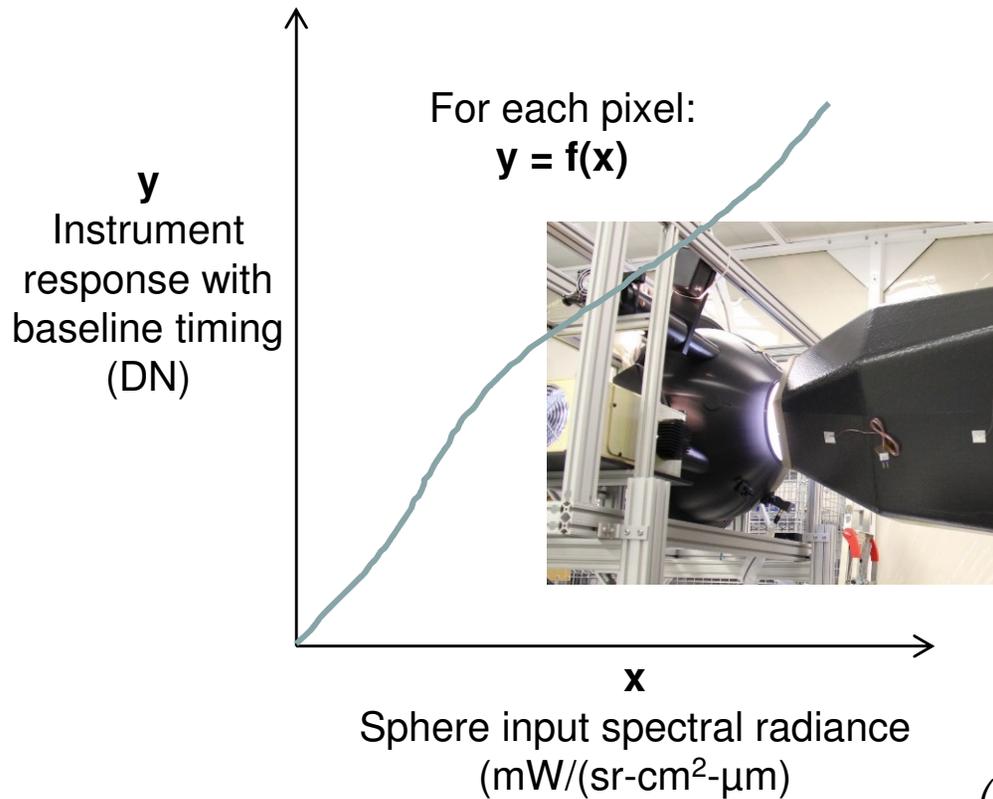




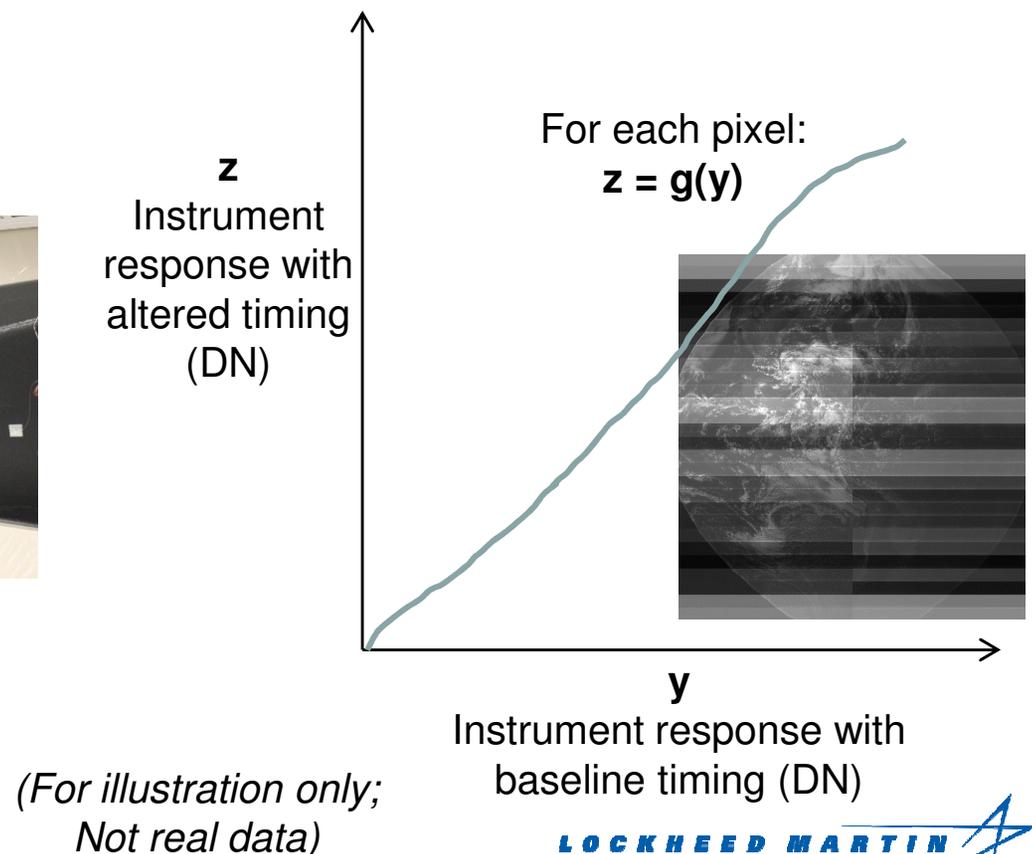
# On-Orbit Calibration

- GLM flight software contains a register sweep macro that can “turn knobs” for the on-board camera timing, without ground commanding
- PLT tool extracts a “delta” calibration:  $z = g(f(x))$

## Ground Calibration



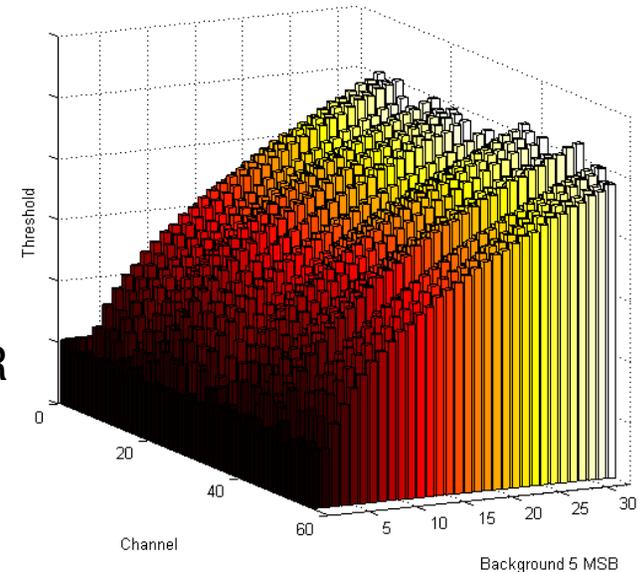
## PLT On-Orbit Calibration





# RTEP Detection Thresholds

- RTEP = Real Time Event Processor
  - Picks out any positive change in the value of a pixel that exceeds a programmable detection threshold, effectively compressing raw video by ~2000x. The RTEP discards data, so must be tuned carefully!
- RTEPs are like lawn mowers
  - Blade too high → low sensitivity to dim lightning
  - Blade too low → jam the downlink with false events
- 56 channels \* 32 BG levels = 1792 thresholds
  - TNR or Threshold-to-Noise Ratio is related to the FER
  - For each threshold,  $TNR = \sqrt{2} \operatorname{erfc}^{-1}(2 \cdot FER / (503))$
  - Initial TNR ~ 5, with a target after RTEP tuning ~4.3
- For RTEP tuning, lightning is a distraction!
  - Can use GPAs to collect only the noise events, and throw out lightning

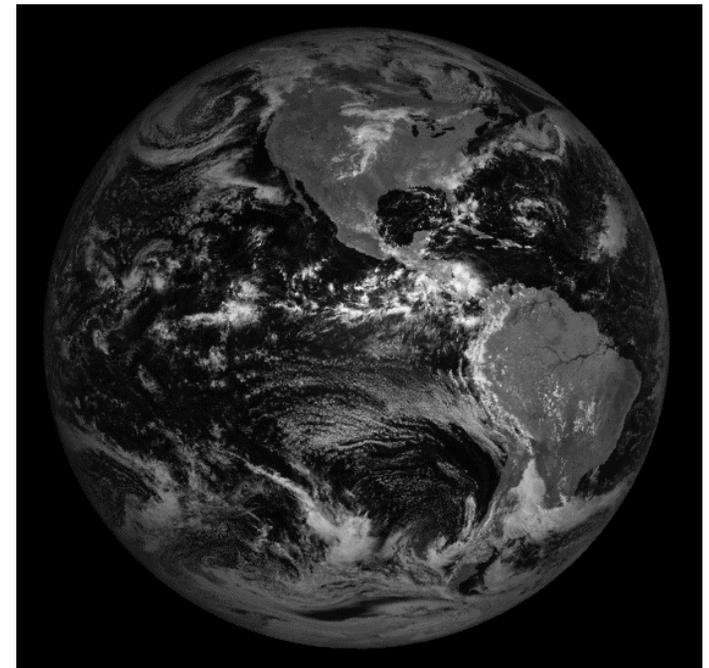


*Example Threshold Table*



# Coastline ID and Navigation

- GLM reports events in pixel coordinates (x,y)
- Need to translate (x,y) to (longitude, latitude) for L1b lightning data
- Coastline ID is a ground image processing algorithm applied to daytime background images from GLM that identifies coastlines and matches them to known database
  - Result is a boresight offset measurement
- Before PLPT and independent navigation assessment, GLM will perform sanity check on navigation GPA parameters that were determined in ground testing
  - Optical distortion map
  - Temperature sensitivities



*Earth viewed at 780 nm by DSCOVR / EPIC*



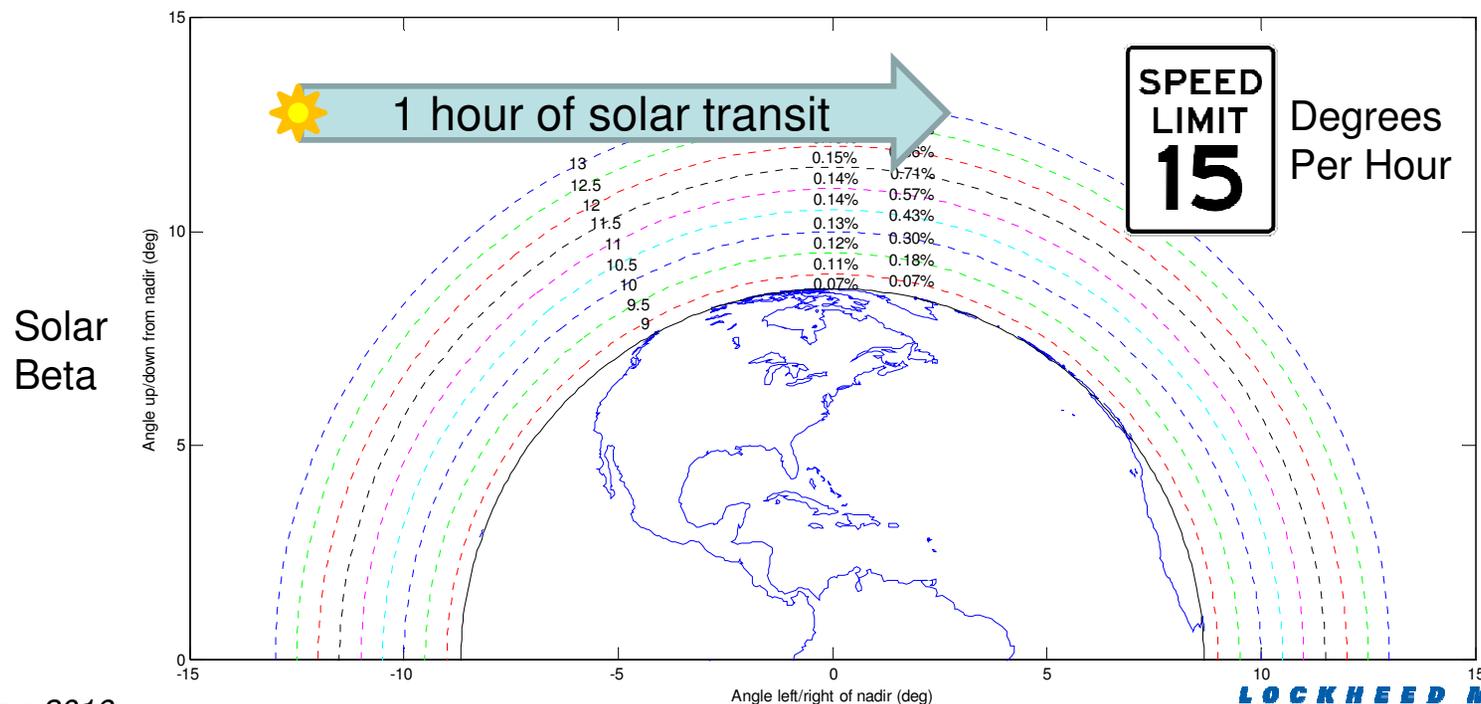
# GPA Filter Tuning

- Ground Processing Algorithms delivered to NASA by GLM vendor
- GPAs are implemented into software by the GS vendor, Harris Corp.
- Ground Processing Algorithm event filters sift out lightning events from other events due to noise, radiation, etc.
  - All event data processing is done on the ground except (by necessity, due to downlink limit) the event detection process.
  - The GPA filters discard data, so must be tuned carefully!
- PLT is performed in Palo Alto on GLM vendor's GPA implementation
  - Does not use GS operational or development environments
  - Requires tight coordination / validation to ensure agreement between GS vendor implementation and GLM vendor implementation
- Matlab PLT tool allows slicing and displaying event data in time and space
  - Regions of interest, selection of filters, etc.
  - This is a Big Data problem (many gigabytes of data per day)



# Solar Intrusion Assessment

- The solar disk is ~50000x brighter than a 100% albedo cloud
- During eclipse season, direct solar illumination enters the GLM optics and reaches almost all the way to the focal plane
- The GLM optics are designed to reject (absorb) this direct solar illumination, using geometrical features and coatings
- We will assess impact to lightning detection within the extent of ZRDQ (Zones of Reduced Data Quality). Data quality product is in development.





# PLT Outputs and PLPT Handoff

- The output of PLT is a parameter file
  - CDRL079 Calibration Data Books, containing an HDF5 file with the thousands of on-board and ground processing parameters
- PLT may also result in changes to algorithms, or new algorithms
  - This is a new instrument, and current GPAs are only our highly educated guess for what will work best to detect lightning
- In practice, PLPT will overlap with PLT
  - Requires incorporation of PLT results into GS development environment, since PLPT will rely on GS L1b product

```
HDF5 GLM_FM1.h5
GLM_CalibrationParameters
|-- EventFilters
|   |-- 2ndLevelThreshold
|   |-- Burst
|   |-- Coherency
|   |-- ContrastLeakage
|   |-- Crosstalk
|   |-- FrameTransfer
|   |-- Glint
|   |-- Mask
|   |-- Overshoot
|   \-- Radiation
|-- INR
|   |-- Coastline
|   |-- Navigation
|   \-- UniversalParameters
|-- InstrumentParameters
|-- RadiometricCalibration
\-- SpacecraftParameters
```

